



## Stress intensity factor for small embedded cracks in weldments

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**ABSTRACT.** In the present work, the stress intensity factor (SIF) of embedded small cracks placed at the weld toe is calculated by means of two procedures based on the Oore-Burns integral. In the first approach, the defect is considered as a circular disk and the SIF is evaluated by means of the Oore-Burns weight function. By taking advantage of a suitable change of variable, the singularity of the weight function on the crack border can be removed. In this way, the numerical evaluation of the SIF is possible without the use of specific integration algorithms, although the nominal stress field becomes singular when the crack approaches a V-sharpe notch. As an example, the obtained equations are applied to a defect located in the neighbourhood of a weld toe with an opening angle of 135 degrees under mode I loading. Subsequently, for a crack similar to a star domain with a border expressed by means of the Fourier series, the SIF is given by means of an explicit equation based on the Oore-Burns weight function.

**KEYWORDS.** Weight function; Stress intensity factor; Three-dimensional crack; Weld.

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## INTRODUCTION

The safety factors in the presence of a defect can be verified by the knowledge of the stress intensity factor (SIF) evaluated for an elastic linear material [1]. As proposed by Bueckner [2] and Rice [3], the SIF calculation can be performed by means of the weight function technique. The numerical analysis is reduced to an evaluation of an integral exactly over the region covered by the crack. This method can be applied both for two-dimensional models (crack as a line) or for three-dimensional problems (crack as a surface). In the literature for two-dimensional cracks, there are many formulations in closed form obtained with different methodologies [4]. However, for the three-dimensional cracks, the number of proposed solutions is poor (see, for example, bibliographical references [5–6]). In fact, in many cases the use of



simplifications in the form of a defect is common and it is comparable to relatively simple geometric shapes of which the solution is already known. Usually, by simplifying, the defect is compared to an elliptical equivalent crack [7], while in order to simulate the propagation phase, the defect is continuously assimilated to a succession of ellipses [8–9].

The weight function introduced by Oore-Burns (OB) allows the value of the SIF of three-dimensional cracks under mode I loading with a generic shape to be calculated [10]. Although not exact, this weight function, for the three-dimensional case [11], is an excellent approximation that retains all the typical characteristics of the weighted functions in [12]. In particular, in the case of a semi-axial ellipse (1, b), when eccentricity  $e$  tends to zero, the main contribution of the Oore-Burns integral differs from Irwin's analytical solution [13] for an amount equal to  $\frac{e^2}{20\sqrt{\pi}}$  [14]. The advantage offered by the Oore-Burns

integral is to use a generic shape of the defect and the evaluation of SIF at each point of the crack contour without the need to consider the elliptical shape. Unfortunately, the difficulties associated with the analytical calculation of the integral result in the user choosing a particular grid [12, 15]. The resolution strategy proposed in [15] is not always easy to reach the convergence condition. However, it should be emphasised that the Oore-Burns integral is recommended by ASM standards in the case of isolated three-dimensional cracks to have an estimate of the SIF value throughout the profile of a real crack. The designer is therefore motivated to find a solution in closed form for a rapid evaluation of the SIF of a crack with generic form [16].

In this paper, we first summarise an analytical technique to get the Oore-Burns integral as an exact solution. Then, for a circular crack, a simple general equation for the SIF estimation is proposed and, as an example, it is used in the case of welded structures. Finally, the Oore-Burns integral is very well approximated by means of an explicit form for an irregular crack shape similar to a star domain under uniform tensile loading.

## ANALYTICAL BACKGROUND

**L**et  $\Omega$  be an open bounded simply connected subset of the plane as in Fig. 1. We define:

$$f(Q) = \int_{\partial\Omega} \frac{ds}{|Q - P(s)|^2} \quad (1)$$

where  $Q = Q(x, y) \in \Omega$ ,  $s$  is the arch-length parameter and the point  $P(s)$  runs over the boundary  $\partial\Omega$ . In their famous work in 1980, Oore-Burns proposed the following expression for the mode I stress intensity factor for a crack subjected to a nominal tensile loading  $\sigma_n(Q)$  evaluated without the presence of the crack:

$$K_I(Q') = \frac{\sqrt{2}}{\pi} \int_{\Omega} \frac{\sigma_n(Q)}{\sqrt{f(Q)} |Q - Q'|^2} d\Omega \quad , \quad Q' \in \partial\Omega \quad (2)$$

Under reasonable hypothesis on the function  $\sigma_n(Q)$ , the integral (2) is convergent and the proof is based on the asymptotic behaviour of  $f(Q)$  [17].

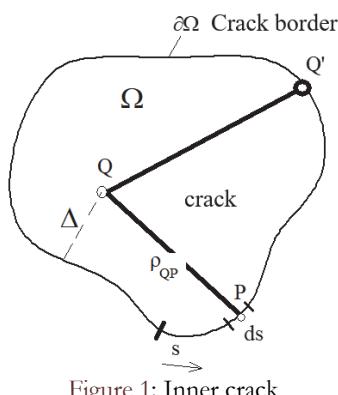


Figure 1: Inner crack

## ORE-BURNS ON A CIRCULAR CRACK

**L**e $\Omega$  be a fixed set. We can reconstruct the OB integral along the front crack  $\partial(\lambda\Omega)$  by its values along  $\partial\Omega$  by the equation

$$K_I(\lambda, \mathcal{Q}', \sigma_n) = \sqrt{\lambda} K_I(1, \mathcal{Q}'/\lambda, \sigma) \quad (3)$$

where  $\sigma = \sigma_n(\lambda\mathcal{Q})$  with  $\lambda > 0$

If we “read” the boundary point  $\mathcal{Q}'$  in terms of an angle  $\alpha$  that is, for example  $\Omega$  is star shaped with respect to the origin, (3) takes the simplest form

$$K_I(\lambda, \alpha, \sigma_n) = \sqrt{\lambda} K_I(1, \alpha, \sigma) \quad (4)$$

In the particular case when  $\Omega$  is a disk of radius  $\bar{a}$  centred at the origin of the plane  $(\bar{x}, \bar{y})$ , we denote by  $(x, y)$  the system in dimensionless coordinates  $x = \bar{x}/\bar{a}$  and  $y = \bar{y}/\bar{a}$  (see Fig. 2).

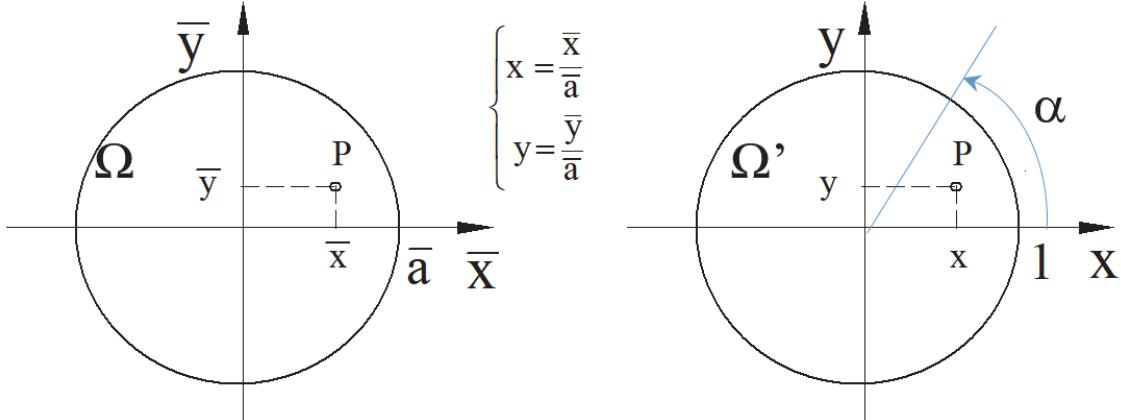


Figure 2: reference circular crack

By definition, for the unit disk  $\Omega'$ , by (1) and (2) it follows:

$$K_{I,0}(\alpha) = \frac{\sqrt{2\bar{a}}}{\pi} \int_{x^2+y^2 \leq 1} \frac{\sigma(x, y) b(x, y)}{(x - \cos \alpha)^2 + (y - \sin \alpha)^2} dx dy \quad (5)$$

With

$$b(x, y) = 1/\sqrt{f(x, y)} \quad (6)$$

$$f(x, y) = \int_{-\pi}^{\pi} \frac{d\theta}{(x - \cos \theta)^2 + (y - \sin \theta)^2} \quad (7)$$

By introducing the change of variables ( $\vartheta \in [0, \pi]$ ,  $\phi \in [0, \pi/2]$ )

$$\begin{cases} x = (1 - r \sin \vartheta) \cos \alpha - r \cos \vartheta \sin \alpha \\ y = r \cos \vartheta \cos \alpha + (1 - r \sin \vartheta) \sin \alpha \end{cases} \quad (8)$$

where



$$r = 2 \sin \vartheta \sin^2 \phi \quad (9)$$

by some simple calculations

$$dx dy = 4 r \sin \vartheta \sin \phi \cos \phi d\vartheta d\phi \quad (10)$$

$$\sqrt{1 - \rho^2} = \sqrt{r} \sqrt{2 \sin \vartheta - r} \quad (11)$$

By using Eqn. (8–11), we have the following expression for the stress intensity factors on the unitary disk

$$K_{I,0}(\alpha) = \frac{4}{\pi \sqrt{\pi}} \int \sigma(x, y) \sin \vartheta \cos^2 \phi d\vartheta d\phi \quad (12)$$

where the integral is computed on the “longitude”  $\vartheta \in [0, \pi]$  and the “latitude”  $\phi \in [0, \pi/2]$ .

Moreover, the pressure  $\sigma(x, y)$  is “read” in the new coordinates  $(\vartheta, \phi)$  for any fixed  $\alpha$ , in the sense that  $x$  and  $y$  are given by (8), with  $r$  being defined by (9). If the crack has a radius equal to  $\bar{a}$  the stress intensity factor becomes

$$K_{I,0}(\alpha) = \frac{4\sqrt{\bar{a}}}{\pi \sqrt{\pi}} \int \sigma(x, y) \sin \vartheta \cos^2 \phi d\vartheta d\phi \quad (13)$$

When  $(\sigma \equiv 1)$ , from (13) we obtain the well-known result:

$$K_{I,0}(\alpha) = \frac{2\sqrt{\bar{a}}}{\sqrt{\pi}} \approx 1.12837\sqrt{\bar{a}} \quad (14)$$

We may test the efficiency of Eqn. (13) by comparison with the special cases of nominal stress distribution considered in the literature [18]. Furthermore, many other new examples have been obtained in reference [19] by changing the shape of the nominal stress  $\sigma$ .

## SIF FOR AN IRREGULAR CRACK SHAPE LIKE A STAR DOMAIN

**F**or a crack like a star domain as reported in Fig. 3, the Oore-Burns integral can be evaluated without a particular numerical procedure. The Oore-Burns integral will be approximated by means of Riemann sums.

Let us use the Cartesian reference system  $x, y$ .  $Q'$  is a point of coordinate  $(R, \alpha)$  on  $\partial\Omega$ . Now we consider a new orthogonal reference system  $u, v$  with origin in  $Q'$  with  $n$  tangent to  $\partial\Omega$  (see Fig. 3). A mesh of size  $\delta$  on  $\partial\Omega$  can be considered, where  $\delta$  divides the length of  $\partial\Omega$ .  $Q_{jk}$  of coordinate  $(k\delta, j\delta)$  in the  $u, v$  plain, and also  $Q_{jk} = R\delta(\cos(j\delta), \sin(j\delta))$ .  $P_m$  in Fig. 3 is a point of coordinate  $m\delta$  with respect to the initial point  $P_0$ . The Riemann sums  $K_I(\delta, Q')$  is given by:

$$K_I(\delta, Q') = \frac{\sqrt{2\delta}}{\pi} \sum_k \frac{\sigma(Q_{jk}) A_{ij}}{k} \quad (15)$$

where

$$A_{ij} = \left( \sum_m |Q_{jk} - P_m|^{-2} \right)^{-\frac{1}{2}} \quad (16)$$

The sum (15) is made on  $0 \leq m \leq \frac{2\pi}{\delta} - 1$ ,  $Q_{jk} \in \Omega$ .

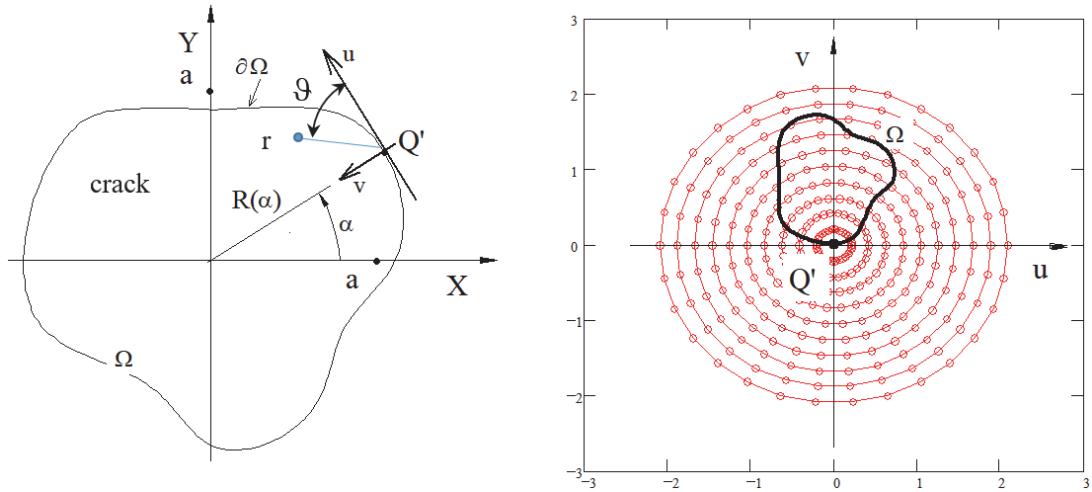


Figure 3: Mesh for Riemann sums

In a previous work [20], we showed that the stress intensity factor can be expressed in the following form:

$$K_I(Q') = K_I(\delta, Q') + \sigma(Q')\sqrt{\delta}C + O(\delta) \quad (17)$$

where

$$C = \frac{\sqrt{2\pi}}{3} \left( \frac{2}{3} - J \right) - I \frac{\sqrt{2}}{\frac{3}{\pi^2}} \zeta \left( \frac{1}{2} \right) \quad (18)$$

with  $I = \int_0^\pi \sqrt{\sin \theta} d\theta$ ,  $J = \int_0^1 \sqrt{\theta \ln \theta} d\theta$  and  $\zeta \left( \frac{1}{2} \right)$  is the well-known Riemann function evaluated in 0.5 ( $C=0.930$ ).

If we know the polar equation of the crack border  $R(\theta)$ , it is convenient to discretise the angle  $\theta$  instead of the arc length  $s$ . Now we consider:

$$\gamma(\theta) = \sqrt{R^2(\theta) + R'^2(\theta)} \quad (19)$$

The coefficients (16) assume the form

$$B_{jk} = \left[ \sum |Q_{jk} - P_m|^{-2} \gamma(m\delta) \right]^{-1} \quad (20)$$

where  $P_m$  is the point of the contour corresponding to  $m \cdot \delta$ . The coefficient  $C$  is then replaced by:

$$D = 0.889 + 0.038 \gamma(\alpha)^{\frac{3}{2}} \cos \left( \frac{2\pi}{\gamma(\alpha)} \right) \quad (21)$$

where  $\alpha$  indicates the position of  $Q'$  for a given origin on the crack border.

The condition  $Q \in \Omega$  can be given in terms of a Fourier series as a function of  $R = R(\theta)$ , so that:

$$R(\theta) = \sum_0^{\infty} (A_r \cos r\theta + B_r \sin r\theta) \quad (22)$$



By setting  $\rho = \sqrt{x^2 + y^2}$ ,  $w = x + iy$ ,  $\mathcal{Q} \in \Omega$  is equivalent to:

$$\rho \leq A_0 + \sum_1^{\infty} \frac{1}{\rho^r} \left( A_r \operatorname{Re} w^r + B_r \operatorname{Im} w^r \right) \quad (23)$$

For example, if  $R(\theta) = 1 - A \cos \theta$  with  $0 \leq A \leq \frac{1}{2}$ , the condition  $\mathcal{Q} \in \Omega$  becomes

$$\rho^2 - \rho - Ax < 0 \quad (24)$$

Finally, the equation for SIF assessments can be rewritten as:

$$K_I(\mathcal{Q}) = K_I(\delta, \mathcal{Q}) + \sigma(\mathcal{Q}) \sqrt{\delta} D + O(\delta) \quad (25)$$

with

$$K_I(\delta, \mathcal{Q}) = \frac{\sqrt{2\delta}}{\pi} \sum_k \frac{\sigma(\mathcal{Q}_{jk}) B_{ij}}{k} \quad (26)$$

(for more details see reference [21]).

## NUMERICAL EXAMPLE

In order to verify the accuracy of the proposed procedure, now we analyse two reference cases: the first is a circular defect at the weld toe, the second is an irregular crack under uniform tensile loading.

Fig. 4 shows a welded T-joint under tensile nominal stress. In Fig. 4, a disc is put at the weld toe of the welded T-joint and the stress intensity factor is evaluated by considering the asymptotic stress field as reported in [22–23]. The crack lies along the bisector at variable distance  $d$  from the weld corner. The T-joint is subjected to a tensile nominal stress, but along the bisector of the weld corner the hoop stress assumes the simple form:

$$\sigma_g = 0.399 K_N r^{-0.326} \quad (27)$$

where  $K_N$  is the Notch Stress Intensity Factors of mode I and it is equal to  $2.46 \text{ MPa mm}^{0.326}$  for a nominal tensile stress  $\sigma_n$  of 1 MPa. This value was calculated by means of a careful notch stress analysis by considering the T-joints of Fig. 5 as three-dimensional components without taking into account the effects of the width as analysed in [24].

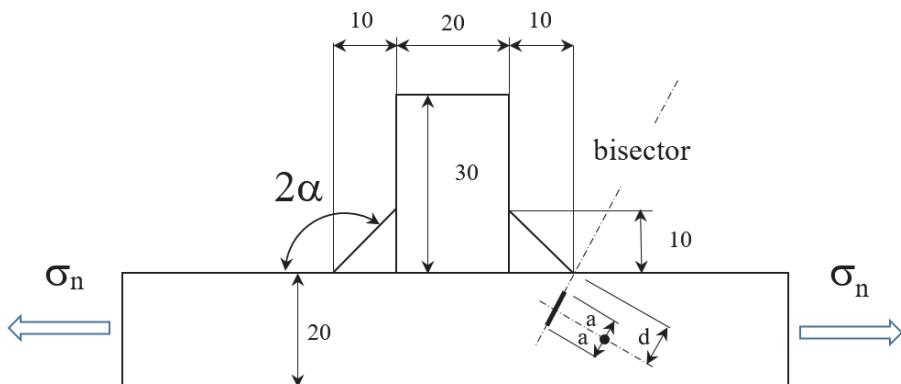


Figure 4: Welded T-joint under tensile loading.

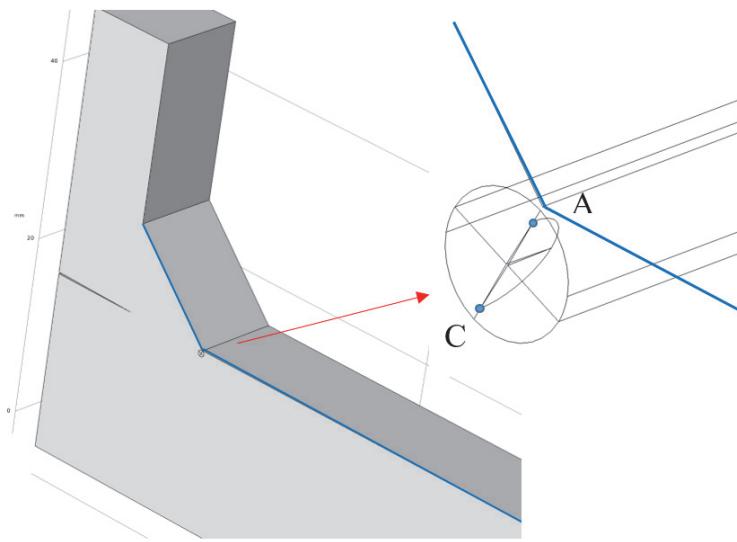


Figure 5: Circular crack at the weld toe of Fig. 4.

Tab. 1 shows the comparison on the Stress Intensity Factor prediction by means of Eqn. (13) and the results obtained with FE analysis. An accurate mesh is considered and the strand of the principal stress along the crack direction is shown in Fig. 6. The dimensions of smaller elements at the tip of the crack were in the order of  $10^{-5}$  mm. Fig. 7, as an example, shows the mesh used for point C.

The difference between the analytical and numerical results is less than 7%. So that, from an engineering point of view, the simple Eqn. (13) can be used for SIF assessments of a small defect in the neighbourhood of the weld toe as made in references [8–9] for semi-circular cracks.

If the crack has an irregular shape not approximable to a disc, Eqn. (25) can be used. In order to estimate the fatigue limit for small cracks or defects under a uniform tensile stress, Murakami and Endo [25–27] proposed  $\sqrt[4]{\text{area}}$  as an empirical parameter to estimate the maximum SIF value of a convex-shaped crack. On the basis of numerous numerical analyses, Murakami and Nemat-Nasser proposed a synthesis relation for the maximum value of the SIF of superficial cracks in the form:  $K_{I,\max} = Y \sigma \sqrt{\pi \sqrt{\text{area}}}$ , where Y is a dimensionless factor that assumes the value of 0.629 for lateral cracks [25]. In the case of internal cracks, the value of Y is approximated to 0.5 [25].

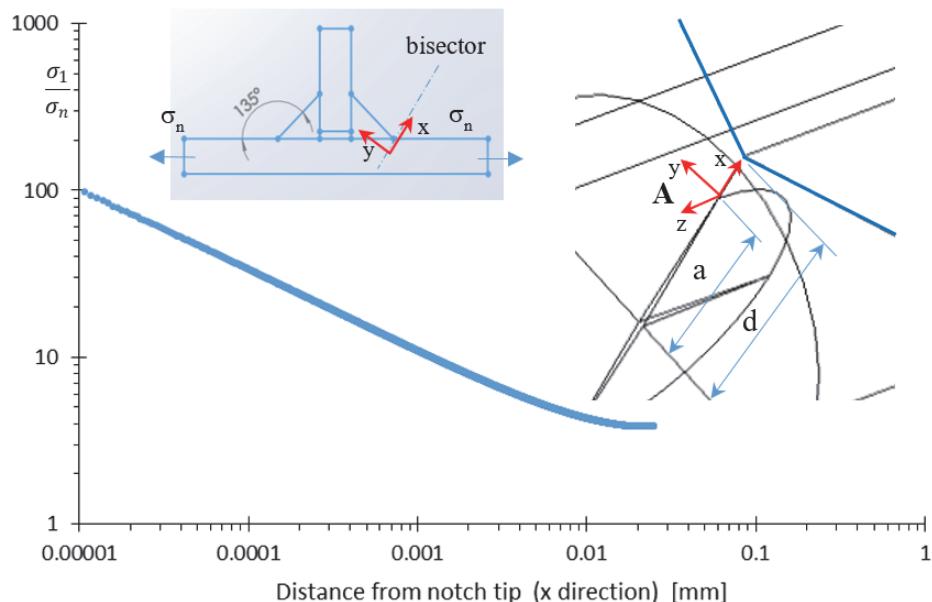


Figure 6: Principal stress  $\sigma_1$  along the bisector for point A of Fig. 4 ( $a=0.1$  mm;  $d=0.15$  mm, opening angle  $2\alpha=135^\circ$ ).

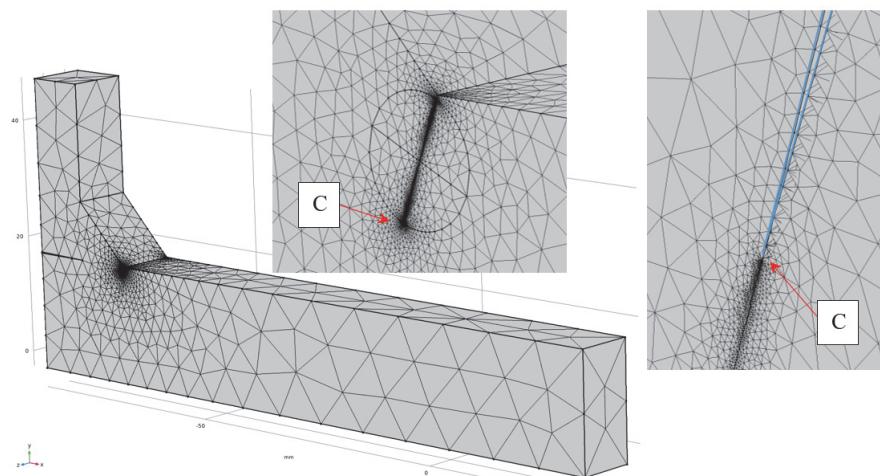


Figure 7: Example of mesh used in the numerical FE analysis for the evaluation of SIF for point C (the dimensions of smaller elements at the tip of the crack were in the order of  $10^{-5}$  mm-  $a=0.3$  mm,  $d=0.4$  mm).

a [mm]	d [mm]	A	C	A	C
		$\frac{K_I}{\sigma_n}$	$\frac{K_I}{\sigma_n}$	$\frac{K_I}{\sigma_n}$	$\frac{K_I}{\sigma_n}$
		Eqn. (13) [mm $^{0.5}$ ]	Eqn. (13) [mm $^{0.5}$ ]	FE [mm $^{0.5}$ ]	FE [mm $^{0.5}$ ]
0.3	0.4	1.088	0.795	1.166	0.816
0.1	0.15	0.818	0.583	0.858	0.611
0.1	0.3	0.566	0.488	0.608	0.524

Table 1: Stress intensity factor at points A and C of Fig. 5 for a nominal tensile loading  $\sigma_n$  (opening angle  $135^\circ$ ).

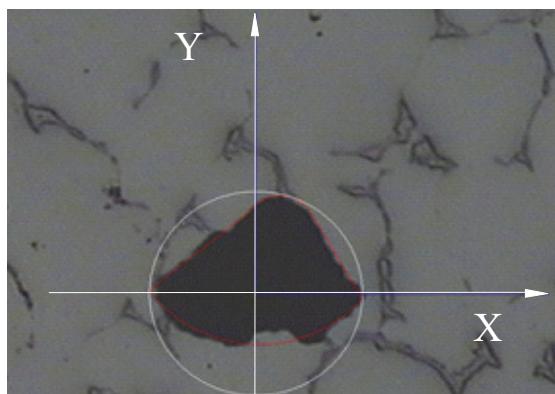


Figure 8: Defects (maximum diameter 0.42 mm)

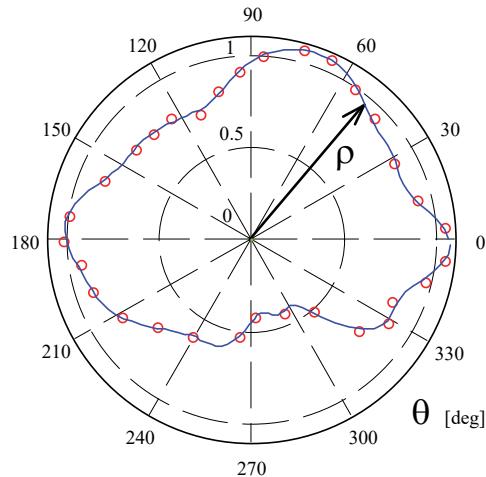


Figure 9: Defect in dimensionless form obtained by dividing the radius by half the value of the maximum diameter.

As an example, the calculation procedure described in the previous section is used to calculate the form factor  $Y$ , which refers to a small defect detected inside a metallic material. Fig. 8 reports an irregular defect of a maximum diameter equal to

0.42 mm. The dimensionless shape of the defects is reported in Fig. 9, where the half maximum diameter is considered in the dimensionless operation.

The numerical results are shown in Fig. 10. The symbols represent the numerical values obtained by means of Eqn. (25), whereas the continuous line is the SIF calculated by considering the defect of Fig. 8 as a first order deviation of a circle (see reference [28] for details). In this case after the reading of the border in the Fourier series, the stress intensity factors are given as a sum of harmonic terms weighed by numerical coefficients derived from the Oore-Burns integral in a similar way to that obtained in references [29–30].

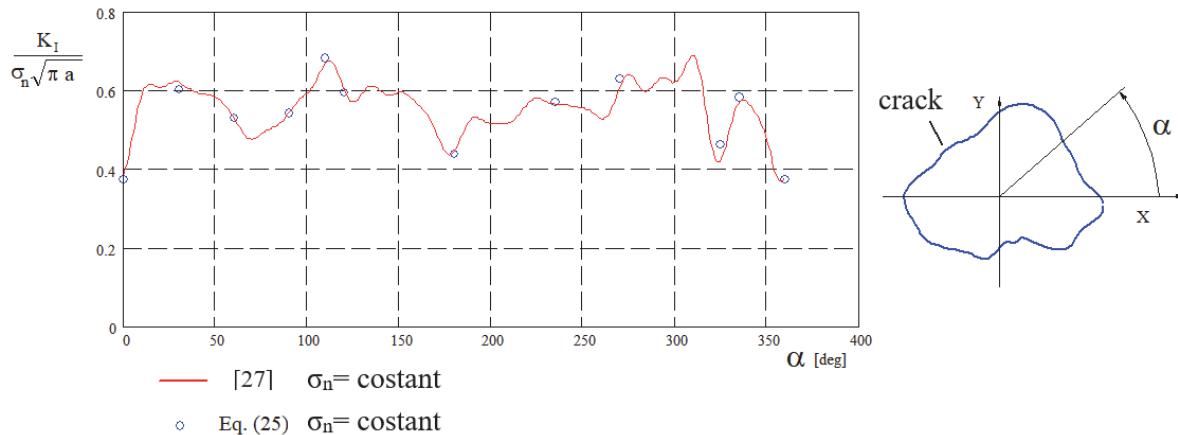


Figure 10: SIF of the crack of Fig. 8 under uniform tensile loading  $\sigma_n$ . ( $a$  is half maximum diameter).

## CONCLUSIONS

The calculation of the stress intensity factor (SIF) of small circular defects located at the weld toe is possible without the use of specific numerical procedures despite the singularity of the stress field when the distance from the notch tip becomes zero. If the defects do not have a circular form, by taking advantage of the Oore-Burns weight function, a procedure is developed for cracks similar to a star domain. In this way, a more precise evaluation of the SIF is possible and the shape factor of non-convex defects should be estimated. The influence of mode II will be take into account in the future.

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